

Using Quadratic Indexes in the Synthesis of Harmonic Disturbance Attenuation Compensators

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Abstract - The paper addresses the synthesis of fixed-structure compensators to attenuate an harmonic disturbance whose frequency is unknown but belongs to a given interval. Besides jointly employing the sensitivity function and Liénard-Chipard Criterion to give guaranteed stability margins, the main novelty of the proposed method lies in using (integral) quadratic indexes of unit step tracking error to shape time-domain performances. In such a way the compensator synthesis is converted into a new minimax optimization problem for which an approximate solution is attainable through known numerical algorithms. The method is applied to a problem arising in the control of wood contouring machines.

I. INTRODUCTION

In the context of linear time-invariant continuous systems consider the standard unity-feedback configuration of Fig. 1 where $P(s)$ and $C(s)$ are respectively the given rational transfer function of the plant and the compensator transfer function to be designed. Signals y , r , e , and d are respectively the controlled output, the reference, the tracking error, and the harmonic disturbance modelled by a single sinusoid whose frequency is unknown but remains in the known range $[\omega^-, \omega^+]$.

Briefly stated the addressed problem is the following. Given any plant $P(s)$ (stable/unstable, minimum/non-minimum phase) design a fixed-structure compensator $C(s)$ such that: (i) the harmonic disturbance is asymptotically attenuated as much as possible for any frequency in the range $[\omega^-, \omega^+]$; (ii) closed-loop stability is achieved with guaranteed stability margins; (iii) time-domain performance specifications are given in terms of constraints on overshoot (M_p), rise time (T_r), and settling time (T_s); (iv) the velocity error constant $K_v > 0$ is assigned (type-one feedback system).

Specific motivation to this problem can be found in the control of wood contouring machines where fixed-structure compensators—such as PD controllers and low-order compensators—can be preferred for implementation easiness [1],[2].

Apart from trial-and-error design based on intensive computer simulations, which is not viable even for moderately complex compensators (such as those with three-four design parameters), a pertinent available design methodology is H_∞ control, particularly suitable to include robustness specifications [3]. However, H_∞

design may not yield the best result when performances involve overshoot, rise time, and settling time; moreover, the designed compensators may be of unnecessary high order [4].

In this paper it is shown how to resort to a minimax optimization problem to solve the posed synthesis problem. Instrumental to the optimization problem formulation is the substitution of time-domain specifications usually given as $M_p \leq \bar{M}_p$, $T_r \leq \bar{T}_r$, $T_s \leq \bar{T}_s$ with a set of constraints on integral quadratic indexes $J_0 \leq \bar{J}_0$, $J_1 \leq \bar{J}_1$, $J_2 \leq \bar{J}_2$. These indexes are defined as $J_i := \int_0^\infty [D^i e(t)]^2 dt$, $i = 0, 1, 2$, where $e(t)$ is the tracking error in response to unit step reference and D^i indicates i th-order derivative. Bar over symbols denote acceptable time-domain performances not to be downgraded.

Closed-form expressions and recursive algorithms to compute indexes J_i are known in the literature [5-10]: the Appendix reports the determinantal method adopted in this work. It has been showed, by qualitative and quantitative reasoning, that imposing constraints on indexes J_i implicitly yields specifications on classical time-domain performances M_p , T_r , and T_s [11].

Paper's organization. Section II exposes the problem analysis which yields to a minimax optimization problem whose approximate solution is obtained through frequency discretization. A worked example, taken from motion control of a tool machine for furniture industry, is exposed in Section III. Conclusions are drawn in last Section IV.

Notation: Denote with $C(k; s)$ the chosen fixed-structure compensator function, rational in all arguments, where $k = (k_1, k_2, \dots, k_p) \in \mathbb{R}^p$ is the design parameter vector. The characteristic polynomial of unity-feedback loop be $\xi(k; s) = a_0(k)s^n + a_1(k)s^{n-1} + \dots + a_n(k)$. The Hurwitz determinant of order i associated to $\xi(k; s)$ is denoted by $\Delta_i(k)$.

The loop transfer function is given by $L(k; s) := C(k; s)P(s)$ and the sensitivity function is $S(k; s) := 1/(1 + L(k; s))$. Denote also with $J_i(k)$, $i = 0, 1, 2$, the rational function of k for which $J_i(k) = J_i$ provided that $\xi(k; s)$ is (Hurwitz) stable. $J_i(k)$ can be computed as shown in the Appendix.

II. FIXED-STRUCTURE COMPENSATOR DESIGN VIA MINIMAX OPTIMIZATION

The design procedure begins by choosing $C(k; s)$ —the fixed-structure compensator transfer function—for which the following assumptions are retained: 1) the

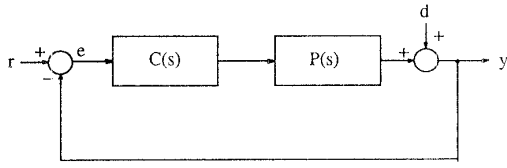


Fig. 1. Unity-feedback control system.

parameter vector k belong to \mathcal{B} , a p -dimensional interval of \mathbb{R}^p ; 2) The first coefficient of $\xi(k; s)$ is always positive : $a_0(k) > 0 \forall k \in \mathcal{B}$; 3) $\lim_{s \rightarrow 0} sL(k; s) = \lim_{s \rightarrow 0} sC(k; s)P(s) = K_v > 0 \forall k \in \mathcal{B}$.

These assumptions are not to be considered restrictive since the interval \mathcal{B} can be enlarged as much as necessary and any fixed-order compensator with prescribed zero-pole pattern (stable/unstable/critical) can always be parametrized to satisfy assumptions 2) and 3) (cf. next Section III).

Remark 1: Note how assumption 3) accounts for the required steady-state accuracy [point (iv) of specifications; cf. Section I].

The measure of asymptotic harmonic attenuation at a given frequency is $|S(k; j\omega)|$. Thus the design objective (i) lies in minimizing, over \mathcal{B} , $\max_{\omega \in [\omega^-, \omega^+]} |S(k; j\omega)|$ which corresponds to posing a minimax synthesis problem.

The following lemma, which is a variant of Liénard-Chipard Criterion [12, pag. 221], is introduced. Its proof can be found in [11].

Lemma 1: Assume $a_0 > 0$. The polynomial $a_0 s^n + a_1 s^{n-1} + \dots + a_{n-1} s + a_n$ is (Hurwitz) stable if and only if:

$$a_r > 0, \Delta_{n-1} > 0, \quad (1a)$$

$$a_{n-1} \geq 0, a_{n-3} \geq 0, \dots, a_l \geq 0 \quad (l \geq 1), \quad (1b)$$

$$\Delta_{n-3} > 0, \Delta_{n-5} > 0, \dots, \Delta_q > 0 \quad (q \geq 2). \quad (1c)$$

Assumption 3) implies that $L(k; s) = \chi(k; s)/[s\psi(k; s)]$ where $\chi(k; s)$ and $\psi(k; s)$ are suitable real polynomials of complex variable s , depending on k , with $\chi(k; 0) \neq 0 \forall k \in \mathcal{B}$. Hence it follows that $\xi(k; s) = s\psi(k; s) + \chi(k; s)$ and the Laplace transform of the unit step tracking error is given by $E(s) = \psi(k; s)/\xi(k; s)$. An immediate consequence of Lemma 1 and formula $J_0(k) = D_0(k)/[a_0(k)a_n(k)\Delta_{n-1}(k)]$ (see formulae 19 and 20 in the Appendix) is the next property which gives necessary and sufficient conditions for closed-loop stability.

Property 2: For a given $k \in \mathcal{B}$, the unity-feedback control system is (Hurwitz) stable if and only if ($l \geq 1; q \geq 2$):

$$a_n(k) \geq 0, \Delta_{n-1}(k) \geq 0, \quad (2a)$$

$$a_{n-1}(k) \geq 0, a_{n-3}(k) \geq 0, \dots, a_l(k) \geq 0, \quad (2b)$$

$$\Delta_{n-3}(k) > 0, \Delta_{n-5}(k) > 0, \dots, \Delta_q(k) > 0, \quad (2c)$$

$$J_0(k) < +\infty. \quad (2d)$$

Stability margins are introduced by imposing that, for a constant real $M > 1$:

$$|1 + L(k; j\omega)| \geq \frac{1}{M} \quad \forall \omega \geq 0. \quad (3)$$

Provided that $\xi(k; s)$ is stable, statement (3), interpreted on the Nyquist plot, implies that gain margin G_M and phase margin P_M satisfy:

$$G_M \geq \frac{M}{M-1}; \quad P_M \geq \cos^{-1} \left(1 - \frac{1}{2M^2} \right). \quad (4a, b)$$

The above expressions (4) constitute guaranteed minimal stability margins. Observe also that (3) is equivalent to

$$|S(k; j\omega)| \leq M \quad \forall \omega \geq 0 \quad \text{or} \quad \|S(k; s)\|_\infty \leq M \quad (5a, b)$$

with $\|\cdot\|_\infty$ denoting the H_∞ -norm of the sensitivity function [3]. When stability margin specifications are assigned as $G_M \geq \bar{G}_M$ and $P_M \geq \bar{P}_M$ the constant M can be chosen according to:

$$\frac{1}{M} \geq \max \left\{ 1 - \frac{1}{\bar{G}_M}, \quad 2 \sin \frac{\bar{P}_M}{2} \right\}. \quad (6)$$

Remark 2: The proposed method for assuring closed-loop stability margins does not permit with certainty to impose $\bar{P}_M \geq 60^\circ$, but this is hardly a truly limitation since 60° for phase margin is fairly high and formula (6) is, in practice, rather conservative.

The set of time-domain performance specifications $T_r \leq \bar{T}_r, T_s \leq \bar{T}_s$, and $M_p \leq \bar{M}_p$ [point (iii) of specifications] is replaced with $J_0 \leq \bar{J}_0, J_1 \leq \bar{J}_1$, and $J_2 \leq \bar{J}_2$ with suitable values \bar{J}_i to be found in the course of synthesizing the actual compensator through iterative use of a minimax optimization solver. The validity of such a substitution has been carefully examined in [11] where a theoretical analysis of time-domain performance relative to a second-order system, or a system with a pair of dominant poles, is exposed. We can add that using these quadratic indexes J_i has been proved effective in all compensator synthesis examined during this research and it is reasonable to expect their usefulness in the broader field of optimization-based control system design. In particular it is emerged the sharp convenience, from the point of view of both the computational burden and time-domain specification effectiveness, of using indexes J_i instead of root clustering regions [13].

The following minimax optimization problem can be considered to design an harmonic disturbance attenuation compensator.

Minimax Optimization Problem: Given finite time-domain performance indexes $\bar{J}_0, \bar{J}_1, \bar{J}_2$, and the upper-bound M on sensitivity H_∞ -norm, find an optimal compensator $C(k^*; s)$ as a solution to ($c > 0; l \geq 1; q \geq 2$):

$$\min_{k \in \mathcal{B}} \max_{\omega \in [\omega^-, \omega^+]} |S(k; j\omega)| \quad (7a)$$

subject to

$$a_n(k) \geq 0, \Delta_{n-1}(k) \geq 0, \quad (7b)$$

$$a_{n-1}(k) \geq 0, a_{n-3}(k) \geq 0, \dots, a_l(k) \geq 0, \quad (7c)$$

$$\Delta_{n-3}(k) \geq \epsilon, \Delta_{n-5}(k) \geq \epsilon, \dots, \Delta_q(k) \geq \epsilon, \quad (7d)$$

$$J_0(k) \leq \bar{J}_0, J_1(k) \leq \bar{J}_1, J_2(k) \leq \bar{J}_2, \quad (7e)$$

$$|S(k; j\omega)| \leq M \quad \forall \omega \geq 0. \quad (7f)$$

This optimization problem, which stems quite naturally from synthesis specifications, is a minimax nonlinear optimization problem with $n+4$ inequality constraints, one of which is the semi-infinite constraint (7f). In particular, note how problem (7) incorporates specifications on closed-loop stability margins through Property 2 and functional inequality (5a).

With regard to inequalities (7d), the positive number ϵ is introduced in order to impose that $\Delta_i(k)$ $i = n-3, n-5, \dots, p$ be strictly positive as required by Property 2. For example, ϵ can be any sufficiently small value compatible with computation precision of the adopted numerical algorithm to solve (7), but in practice, it can always be set to zero. Indeed the following observations hold (cf. [11]):

a) In the case that $n \leq 6$, setting $\epsilon = 0$ inequalities (7b,c,d) with $J_0(k) \leq \bar{J}_0$ if satisfied for some $k \in \mathcal{B}$ assure with certainty closed-loop stability.

b) In case that $n > 6$, setting $\epsilon = 0$ inequalities (7b,c,d) with $J_0(k) \leq \bar{J}_0$ if satisfied for some $k \in \mathcal{B}$ assure with probability one, over the space of all compensator transfer functions $C(k; s)$, closed-loop stability.

Remark 3: In (7) at the optimal solution k^* , the active constraints can only be found among inequalities (7e,f) and/or the boundaries of \mathcal{B} . To know this could be useful in devising an ad hoc algorithmic procedure to solve (7).

Remark 4: The optimization problem (7) can exhibit various “conflicts” among design objectives. In particular it is known as *waterbed effect* [3, pag. 97], for non-minimum phase plant, the contention between minimizing $\max_{\omega \in [\omega^-, \omega^+]} |S(k; j\omega)|$ and satisfying $|S(k; j\omega)| \leq M \quad \forall \omega \geq 0$ with M not too large. In the present context of fixed-structure compensator synthesis, this conflict may even arise for minimum phase plants.

Remark 5: The reason of using stability constraints which are derived from Liénard-Chipard Criterion instead from the well-known Routh-Hurwitz Criterion is that the latter criterion requires computations of all the Hurwitz determinants whereas the former (approximately) requires only half of them.

Exact determination of the optimal compensator $C(k^*; s)$ is a difficult global optimization problem [14]. On the other hand, an approximate solution can be obtained recasting problem (7) into the following discrete minimax problem through a suitable discretization of the frequency axis.

Discrete Minimax Optimization Problem: Choose discretization frequencies in the range $[\omega^-, \omega^+]$ and its complement: $\delta_i \in [\omega^-, \omega^+]$, $i = 1, 2, \dots, m$; $\eta_i \in (0, \omega^-) \cup (\omega^+, +\infty)$ $i = 1, 2, \dots, v$. Find a solution to ($l \geq 1$; $q \geq 2$):

$$\min_{k \in \mathcal{B}} \max_{i=1, \dots, m} \{|S(k; j\delta_i)|\} \quad (8a)$$

subject to

$$a_n(k) \geq 0, \Delta_{n-1}(k) \geq 0, \quad (8b)$$

$$a_{n-1}(k) \geq 0, a_{n-3}(k) \geq 0, \dots, a_l(k) \geq 0, \quad (8c)$$

$$\Delta_{n-3}(k) \geq 0, \Delta_{n-5}(k) \geq 0, \dots, \Delta_q(k) \geq 0, \quad (8d)$$

$$J_0(k) \leq \bar{J}_0, J_1(k) \leq \bar{J}_1, J_2(k) \leq \bar{J}_2, \quad (8e)$$

$$|S(k; j\eta_i)| \leq M \quad (i = 1, \dots, v). \quad (8f)$$

Solution to optimization problem (8) can be attained by means of sequential quadratic programming (SQP) [15,16] for which an implementation is given by the “Matlab Optimization Toolbox” [17].

Caution, however, has to be used with any implementation of a SQP method to solve (8), since these methods can not guarantee convergence to a global solution.

III. AN EXAMPLE

A. The Problem

The electrical drive assigned to axis control of a wood contouring machine is characterized, disregarding high-frequency behaviour, by transfer function

$$P(s) = g_p \frac{\omega_n^2}{s(s^2 + 2\delta\omega_n s + \omega_n^2)} \quad (9)$$

where $g_p = 0.10[\text{ms}^{-1}\text{V}^{-1}]$, $\omega_n = 97$ [rad/s], and $\delta = 0.25$. Depending on tool machine operations this plant is subjected to a sinusoidal disturbance whose frequency is in the range $[\omega^-, \omega^+]$ with $\omega^- = 40$ [rad/s] and $\omega^+ = 80$ [rad/s].

It is required to synthesize a compensator in a unity-feedback control system which satisfies requirements (i)-(iv) (cf. Section I) with

$$K_v = 30[\text{s}^{-1}], \quad (10a)$$

$$G_m \geq 3, P_m \geq 40^\circ, \quad (10b)$$

$$T_r \leq 0.05[\text{s}], T_s \leq 1[\text{s}], M_p \leq 5\%. \quad (10c)$$

B. The Fixed-structure Compensators

1) PD controller

Consider the PD controller (proportional-derivative) given by

$$C_1(k; s) = g_c \left(1 + \frac{k_1 s}{1 + (k_1/N)s} \right) \quad (11)$$

with N the “noise” factor set to 30, k_1 the time constant of the derivative action, and g_c the static gain of the

compensator. Here we simply have $k := k_1 \in \mathcal{B} \subseteq \mathfrak{R}$ and choose $\mathcal{B} := [10^{-6}, 0.06]$.

2) third-order compensator

Consider a third-order biproper compensator

$$C_2(s) = A \frac{(s - z_1)(s - z_2)(s - z_3)}{(s - p_1)(s - p_2)(s - p_3)} \quad (12)$$

with the following pole-zero pattern:

- a) all zeros and poles lie in the open left half-plane;
- b) all zeros and poles are real except two of the zeros which are conjugate complex: let them be z_2 and z_3 .

Then compensator (12) can be parametrized according to

$$C_2(s) = C_2(k; s) = g_c \frac{(1 + k_4 s)(1 + 2k_5 k_6 s + k_6^2 s^2)}{(1 + k_1 s)(1 + k_2 s)(1 + k_3 s)} \quad (13)$$

with corresponding relations:

$$g_c = C_2(0) = (Az_1 z_2 z_3)/(p_1 p_2 p_3), \quad k_i = -(1/p_i) > 0, \quad i = 1, 2, 3, \quad k_4 = -(1/z_1) > 0, \quad k_5 = -(z_2 + z_3)/(2\sqrt{z_2 z_3}), \quad k_6 \in (0, 1), \quad k_6 = 1/\sqrt{z_2 z_3} > 0.$$

The compensator design vector is $k := [k_1 k_2 k_3 k_4 k_5 k_6]^T \in \mathcal{B} \subseteq \mathfrak{R}^6$ with $\mathcal{B} := [10^{-3}, 1] \times [10^{-3}, 1] \times [10^{-3}, 1] \times [10^{-2}, 10] \times [0.1, 0.9] \times [0.005, 1]$.

C. Computational Results

For both the proposed compensators it is readily apparent that all the assumptions made in Section II can be satisfied. Indeed $a_o(k) = k_1/N > 0 \forall k \in \mathcal{B}$ relative to PD controller (11) and $a_o(k) = k_1 k_2 k_3 > 0 \forall k \in \mathcal{B}$ relative to (13). The third assumption $\lim_{s \rightarrow 0} sC(k; s)P(s) = K_v = 30[s^{-1}] \forall k \in \mathcal{B}$ is verified if for both compensators $g_c = 300$ [V/m].

The discrete minimax problem (8) has been set up with $m = 5$ and $v = 20$ and frequency values suitably spaced on \mathfrak{R}^+ . In such a way the resulting problem was solved using the MATLAB optimization toolbox [17]. Specifically the iterative use of the routine `minimax()` has permitted to determine the upper-bounds \bar{J}_0 , \bar{J}_1 , \bar{J}_2 , and M for each synthesized compensator. Results are the following.

1) PD controller

With $\bar{J}_0 = 0.016$, $\bar{J}_1 = 50$, $\bar{J}_2 = 6 \cdot 10^5$, and $M = 1.4$ the obtained (sub-)optimal synthesis is given by $k^* = k_1^* = 0.0153$ with

$$\max_{\omega \in [\omega^-, \omega^+]} |S(k^*; j\omega)| = 0.671, \quad (14a)$$

$$G_m = 3.56, \quad P_m = 40.4^\circ, \quad (14b)$$

$$T_r = 0.0231[s], \quad T_s = 0.286[s], \quad M_p = 2.1\%. \quad (14c)$$

Figures 2 and 3 report respectively the corresponding unit step response and magnitude plot of sensitivity.

2) third-order compensator

With $\bar{J}_0 = 0.10$, $\bar{J}_1 = 100$, $\bar{J}_2 = 7 \cdot 10^7$, and $M = 1.4$ the (sub-)optimal synthesis is given by $k^* = [0.001, 0.001, 0.0091, 0.139, 0.265, 0.00681]^T$ with

$$\max_{\omega \in [\omega^-, \omega^+]} |S(k^*; j\omega)| = 0.193, \quad (15a)$$

$$G_m = 11.6, \quad P_m = 42.7^\circ, \quad (15b)$$

$$T_r = 0.0095[s], \quad T_s = 0.495[s], \quad M_p = 2.6\%. \quad (15c)$$

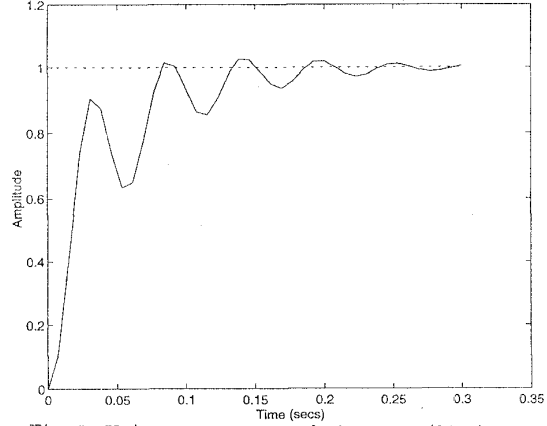


Fig. 2. Unit step response relative to $C_1(k^*; s)$.

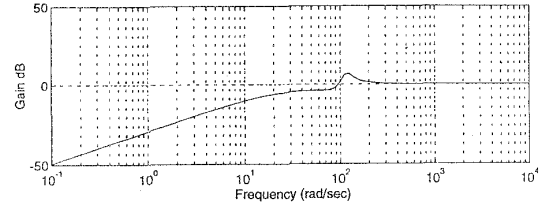


Fig. 3. Magnitude plot of $S(k^*; j\omega)$ relative to $C_1(k^*; s)$.

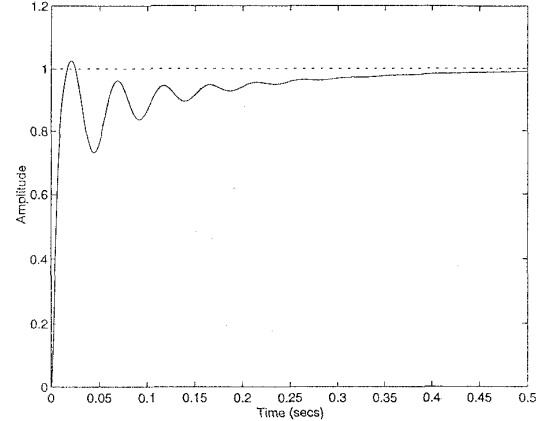


Fig. 4. Unit step response relative to $C_2(k^*; s)$.

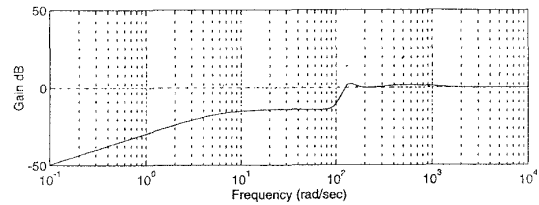


Fig. 5. Magnitude plot of $S(k^*; j\omega)$ relative to $C_2(k^*; s)$.

See also Figures 4 and 5.

Remark 6: Due to the type-one transfer function of the plant the PD controller (11) operates like a PID controller.

Remark 7: In both synthesis the critical constraint is given by the specification on phase margin. Note how this harsh constraint determines in the suboptimal PD controller synthesis a poor harmonic attenuation.

Remark 8: Superiority of sub-optimal synthesis 2) over that of 1) is evident from results (14) and (15). Not surprisingly, as congruent with control system design experience, the sub-optimal compensator $C_2(k^*; s)$ can almost be interpreted as that determined with an optimal zero-assignment procedure (cf. the associated k^*).

IV. CONCLUSIONS

In this paper it has been proposed an optimization method to synthesize (sub-)optimal fixed-structure compensators to attenuate harmonic disturbances over a given frequency range. The method encompasses the use of quadratic indexes of unit step tracking error to satisfy additional time-domain specifications given by upper-bounds on rise time, settling time, and overshoot. The overall procedure, which has been proved effective for small-to-medium synthesis problem, is best to be used in an interactive environment of computer-aided control systems design.

From the viewpoint of optimality and robustness two are the main issues in completing this study: (i) creating ad hoc algorithms to obtain guaranteed global solutions for the synthesis problem; (ii) using an uncertain model of the plant with structured/unstructured perturbations (i.e. robustness beyond gain and phase margins). Future research will address these issues.

APPENDIX

For the reader's convenience this appendix exposes succinctly the determinantal method to compute $J_0 = \int_0^\infty [e(t)]^2 dt$ which is due to Katz [9] and subsequently reported in [10] (see also [6, pp. 8-11]).

Let $E(s)$ denote the Laplace transform of $e(t)$ with

$$E(s) = \frac{B(s)}{A(s)} = \frac{b_1 s^{n-1} + b_2 s^{n-2} + \dots + b_n}{a_0 s^n + a_1 s^{n-1} + \dots + a_n}$$

Assume $a_0 \neq 0$ and $A(s)$ (Hurwitz) stable for which quadratic index J_0 is well defined.

Introduce polynomial $F(s) := (-1)^{n-1} \{f_1 s^{n-1} + f_2 s^{n-2} + \dots + f_n\}$, $f_i \in \mathbb{R}$, $i = 1, \dots, n$ which has to satisfy the polynomial identity:

$$B(s)B(-s) = F(-s)A(-s) + F(s)A(s).$$

This relation implies that f_i must satisfy a linear matrix equation for which, under current assumptions, an unique solution always exists; then, by virtue of Parseval's theorem, it has been proved that $J_0 = f_1/a_0$.

Using Cramer's rule, the following closed-form expression of f_1 can be obtained:

$$f_1 = \frac{\det \begin{pmatrix} g_1 & a_0 & 0 & 0 & \dots & 0 \\ g_2 & a_2 & a_1 & a_0 & \dots & 0 \\ g_3 & a_4 & a_3 & a_2 & \dots & . \\ . & . & . & a_4 & \dots & . \\ . & . & . & . & \dots & . \\ g_n & 0 & . & . & \dots & a_n \end{pmatrix}}{\det \begin{pmatrix} a_1 & a_0 & 0 & 0 & \dots & 0 \\ a_3 & a_2 & a_1 & a_0 & \dots & 0 \\ a_5 & a_4 & a_3 & a_2 & \dots & . \\ . & . & . & a_4 & \dots & . \\ . & . & . & . & \dots & . \\ 0 & 0 & . & . & \dots & a_n \end{pmatrix}}$$

where

$$g_i = \frac{1}{2} \sum_{j=1}^{2i-1} (-1)^{j-1} b_{2i-j} b_j \quad (b_j = 0 \text{ if } j > n). \quad (16)$$

Since the Hurwitz determinant of order n —relative to $A(s)$ —can be expressed as

$$\Delta_n = \det \begin{pmatrix} a_1 & a_3 & a_5 & . & \dots & 0 \\ a_0 & a_2 & a_4 & . & \dots & 0 \\ 0 & a_1 & a_3 & . & \dots & . \\ 0 & a_0 & a_2 & a_4 & \dots & . \\ . & . & . & . & \dots & . \\ 0 & 0 & 0 & . & \dots & a_n \end{pmatrix} \quad (17)$$

it follows that

$$J_0 = \frac{D_0}{a_0 \Delta_n}, \quad (18)$$

with

$$D_0 := \det \begin{pmatrix} g_1 & g_2 & g_3 & . & \dots & g_n \\ a_0 & a_2 & a_4 & . & \dots & 0 \\ 0 & a_1 & a_3 & . & \dots & . \\ 0 & a_0 & a_2 & a_4 & \dots & . \\ . & . & . & . & \dots & . \\ 0 & 0 & 0 & . & \dots & a_n \end{pmatrix} \quad (19)$$

Taking into account that $\Delta_n = a_n \Delta_{n-1}$, it follows an other determinantal expression of J_0 :

$$J_0 = \frac{D_0}{a_0 a_n \Delta_{n-1}} \quad (20)$$

From formulae (18), (17), (19), and (16) it is apparent that J_0 is a rational expression whose arguments are the coefficients of polynomials $A(s)$ and $B(s)$.

The determinantal formula (20) can be easily generalized to compute $J_i = \int_0^\infty [D^i e(t)]^2 dt$.

Indeed it could be proved that

$$J_i = \frac{D_i}{a_0 a_n \Delta_{n-1}}, \quad i = 1, 2, \dots \quad (21)$$

with D_i being a determinant similarly defined as in (19).

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